



GLAST Large Area TelescopeCalorimeter Subsystem 10.0 Assembly & Test

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Calorimeter Assembly Overview

- 18 Identical Calorimeter **Modules**
 - 1 Qualification Model

PEM

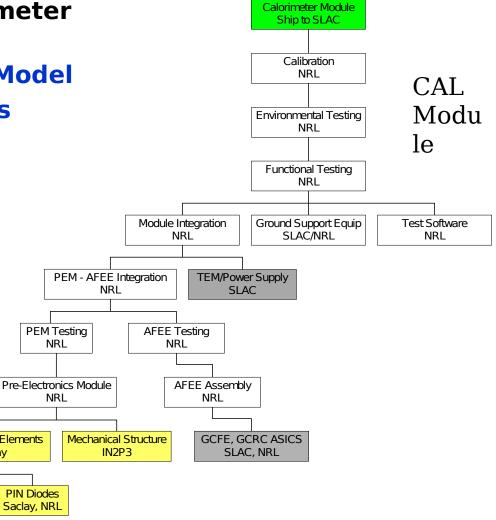
Assy

Csl Crystals Sweden

Csl Detector Elements

Saclay

- **16 Flight Models**
- 1 Flight Spare
- 1st 4 units are LAT **Calibration Unit**





PEM Testina

NRL

NRL

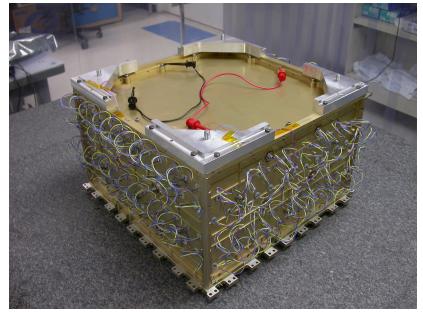
PIN Diodes

Saclay, NRL



PEM Assembly

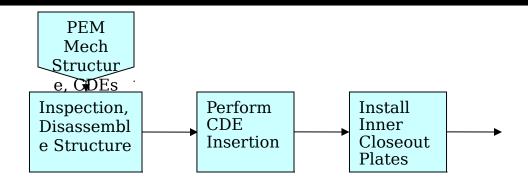
- Pre-Electronic Modules (PEM)
 - Assembly Performed by Naval Research Lab
 - Mechanical Structure, from IN2P3, France
 - Crystal Detector Elements, from CEA, France
- PEM Testing
 - Opportunity to uncover assembly issues prior to next
 - M40h Testing verifies optical performance of CDEs
 - Failed CDEs could be replaced relatively easily at this stage of assembly







PEM Assembly Flow



Goals

- Inspect PEM Mechanical Structure, CDEs verify dimensions
- Assemble Pre-Electronics Module

Inputs

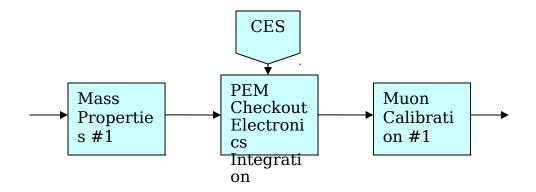
- PEM Mechanical Structure, Data book
- Crystal Detector Elements, Data book
- Elastomeric Cords & Bumper frames
- Special MGSE: CDE Insertion tooling, Closeout Plate Integration tooling

Outputs

- Assembled PEM
- PEM Assembly data book: CDE Identification and insertion locations



PEM Acceptance Tests



Goals

- Inspect PEM, verify dimensions and weight
- Verify that PEM meets light yield and light attenuation specs

☐ Inputs

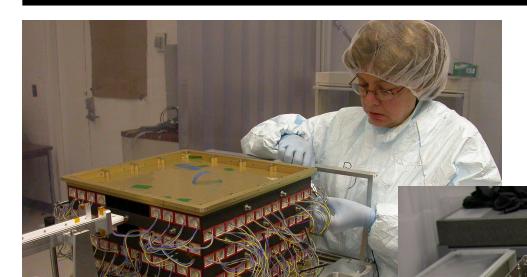
- Assembled PEM
- PEM Data book
- Special EGSE: PEM Checkout Electronics System (lab electronics, DAQ)

Outputs

- Accepted PEM
- CsI light yield and light attenuation maps



PEM Assembly & Test



Engineering Model CDE insertion nearing completion

Engineering Model PEM Muon Calibration test setup





PEM Assembly Schedule

Level 3 Schedule - FM3 & FM4 PEM Assemblies

W B S	A c tivity	A c tivity	To ta I		arly Early	FY03 FY04
	ID	D e s c rip tio n	Float	Dur S	tart Finish	AUG SEP OCT NOV DEC JAN FEB MAR APR MAY JUN JUL AUG SE
CALFM3						
	CTRONICS MODULE (PE					_
4 .1 .5 .6 .2 .3 .5	5 C 6 2 3 0 0 4 0 0	IN: Receive FM 3 Mechanical Struct	102	0	0 9 /1 2 /0 3	■ IN Receive FM 3 Mechanical Struct
4 .1 .5 .6 .2 .3 .5	5 C 6 2 3 0 0 4 1 0	In spect structure	102	1 09/1	5/03 09/15/03	Hnspect structure
4 .1 .5 .6 .2 .3 .5	5 C 6 2 3 0 0 4 2 0	IN: FM3 CDE	6 8	0	1 0 /2 7 /0 3	N: FM3 CDE
4 .1 .5 .6 .2 .3 .5	5 C 6 2 3 0 0 4 3 0	C D E Inspection	6 8	4 10/2	8 / 0 3 1 0 / 3 1 / 0 3	CDE Inspection
4 .1 .5 .6 .2 .3 .5	5 C 6 2 3 0 0 4 4 0	CDE - Mech Struct Integration	6.8	5 11/03	3 / 0 3 1 1 / 0 7 / 0 3	CDE - Mech Struct Integration
4 .1 .5 .6 .2 .3 .5	5 C 6 2 3 0 0 4 5 0	PEM Cell Closeout	6.8	1 11/10	11/10/03	PEM Cell Closeout
4 .1 .5 .6 .2 .3 .5	5 C 6 2 3 0 0 4 6 0	Mass Properties #1	6.8	1 11/11	/03 11/11/03	Mass Properties #1
4 .1 .5 .6 .2 .3 .5	5 C 6 2 3 0 0 4 7 0	PEM Checkout Elect Integration	6.8	1 11/12	2/03 11/12/03	PEM Checkout Elect Integration
4 .1 .5 .6 .2 .3 .5	5 C 6 2 3 0 0 4 8 0	Muon calibration #1	6.8	3 11/13	3 /0 3 11 /1 7 /0 3	Muon calibration #1
4 .1 .5 .6 .2 .3 .5	5 C 6 2 3 0 0 4 9 0	IA: FM 3 PEM Ready	6.8	0	11/17/03	♠ IA: FM3 PEM Ready
CALFM4						
	CTRONICS MODULE (PE	1				
4 .1 .5 .6 .2 .3 .6	5 C 6 2 3 0 0 5 0 0	IN: Receive FM 4 Mechanical Struct	103	0	0 9 /2 6 /0 3	IN: Receive FM 4 Mechanical Struct
4 .1 .5 .6 .2 .3 .6	5 C 6 2 3 0 0 5 1 0	In spect structure	103	1 09/2	9/03 09/29/03	Hnspectstructure Hnspectstructure
4 .1 .5 .6 .2 .3 .6	5 C 6 2 3 0 0 5 2 0	IN: FM 4 CDE	6 9	0	11/10/03	♦ IN: FM4 C D E
4 .1 .5 .6 .2 .3 .6	5 C 6 2 3 0 0 5 3 0	C D E Inspection	6 9	4 11/11	/03 11/14/03	CDE Inspection
4 .1 .5 .6 .2 .3 .6	5 C 6 2 3 0 0 5 4 0	CDE - Mech Struct Integration	6 9	5 11/17	1/03 11/21/03	CDE - Mech Struct Integration
4 .1 .5 .6 .2 .3 .6	5 C 6 2 3 0 0 5 5 0	PEM Cell Closeout	6 9	1 11/24	1/03 11/24/03	IPEM Cell Closeout
4.1.5.6.2.3.6	5 C 6 2 3 0 0 5 6 0	Mass Properties #1	6 9	1 11/25	5/03 11/25/03	M ass Properties #1
4 .1 .5 .6 .2 .3 .6	5 C 6 2 3 0 0 5 7 0	PEM Checkout Elect Integration	6 9	1 11/26	5/03 11/26/03	I-PEM Checkout Elect Integration
4.1.5.6.2.3.6	5 C 6 2 3 0 0 5 8 0	M uon calibration #1	6 9	3 12/0	1/03 12/03/03	Muon calibration #1
4 .1 .5 .6 .2 .3 .6	5 C 6 2 3 0 0 5 9 0	IA: FM 4 PEM Ready	6 9	0	1 2 /0 3 /0 3	→ IA: FM 4 PEM Ready





Calorimeter Module Assembly

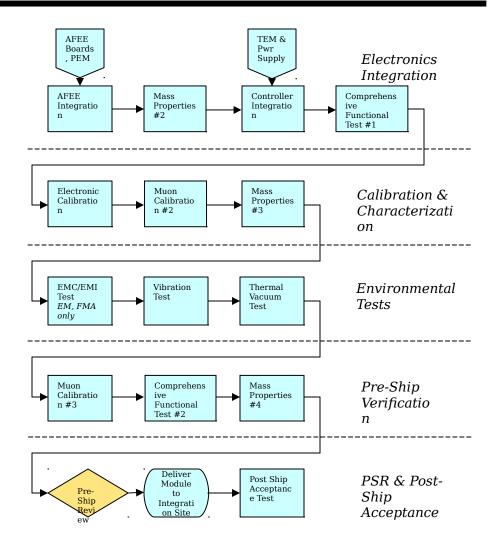
- Assemble Calorimeter Tower Modules
 - Pre-Electronics Module
 - Assembled and Tested at NRL
 - Assemble & Test Analog Front End Elect (AFEE) boards
 - AFEE PWB Design, from SLAC
 - ASIC Designs, from SLAC
 - Tower Electronics Module & Power Supply, from SLAC
- □ Test
 - Functional Testing
 - **Environmental Testing**
 - Calibration
- Deliver to SLAC
 - Perform Acceptance Test



Assembly and Test Flow

Five stages in Assembly & Test sequence:

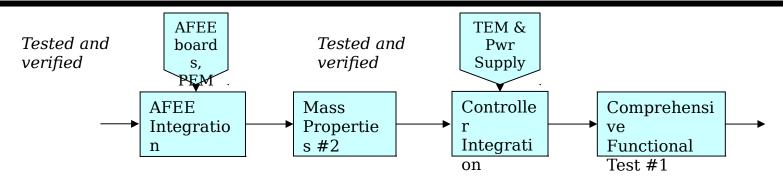
- 1) Electronics integration
- 2) Calibration, baseline
- 3) Environmental tests
- 4) Pre-ship verification
- 5) Delivery & Post-ship Acceptance







Electronics Integration



□ Goals

- Integrate Flight Front-end electronics onto PEM
- Integrate TEM Controller & Power Supply
- Establish weight and physical dimensions of CAL Module
- Establish baseline system performance

Inputs

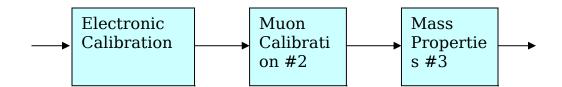
- Verified PEM
- Flight AFEE boards, previously tested and verified
- EM/Flight TEM, previously tested and verified
- EM/Flight Power Supply, previously tested and verified
- Special GSE: Assembly/Rotation Stand, Calorimeter Test Stand (CTS)

Output

Integrated, fully tested CAL Tower Module



Calibration & Characterization



Goals

Establish baseline gain, linearity, and mass properties of integrated module

Inputs

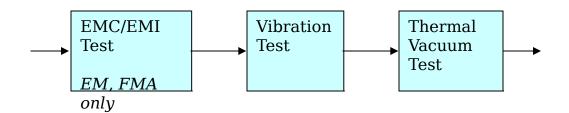
- Integrated and fully tested CAL Tower Module
- Special EGSE: Calorimeter Test Stand

Output

- 768 electronic gain and linearity curves per Module
 - One for each energy range: 96 crystals \times 2 faces \times 4 ranges
- 384 optical gains per Module
 - One for each PIN: 96 crystals \times 2 faces \times 2 PIN Diodes
 - Optical gain is electrons in FE per MeV deposited in xtal
- Mass, dimensions
- Calibrated CAL Module



Environmental Tests



Goals

Ensure Module safety and performance against thermal, pressure,
 vibration, shock, and electromagnetic excursions expected during flight.

Inputs

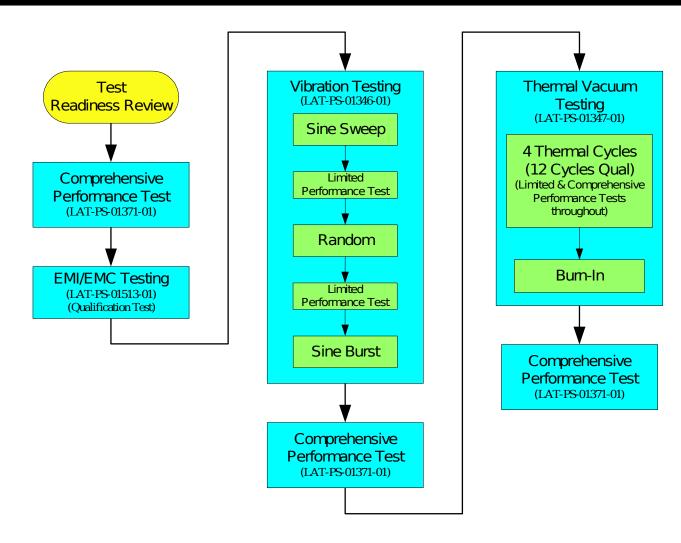
- Fully functional, calibrated CAL Tower Module
- Thermal-Vacuum, Vibration, EMC/EMI facilities at NRL
 - Note test levels vary as appropriate:
 - FMA: Qualification levels, EMC/EMI testing
 - FMB: Acceptance level Vibration, Qual level Thermal Vacuum
 - FM1-FM16: Acceptance levels
- Special GSE: Vibration & Thermal-Vacuum fixtures, Calorimeter Test Stand

Outputs

Environmentally Tested Flight Unit



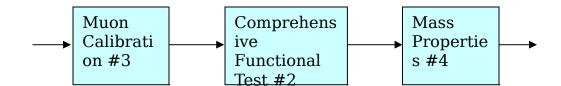
Environmental Test Flow







Post Environmental Verification

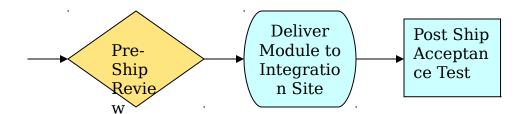


- □ Goals
 - Final verification of Cal Module for delivery
- Inputs
 - Fully Tested CAL Tower Module
 - Special GSE: Calorimeter Test Stand
- Outputs
 - Verified CAL Module ready for Integration
 - Flight Documentation for Pre-Ship Review





Delivery & Acceptance



Pre-Ship Review

- Review Board consists of Subsystem Manager,
 A&T Manager, Systems Engineer, QA Engineer,
 Lead Engineers, others as deemed necessary
- Walk-through A&T flow, review Test Reports,
 Resolution Reports, status of all anomalies, etc
- Flight Unit Delivery
- Post-Ship Acceptance Test
 - Verifies functionality of delivered CAL Module
 - Formal Acceptance by LAT



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Assembly and Test Schedule

Duration of assembly and test phases (working days for each Module)

		Module						
Phase	EM	FMA,FM B	FM1-6	FM7- 16				
Electronics Integration	25	27	12	11				
Calibration	12	6	5	4				
Environmental	44	41	13	13				
Pre-ship Verification	10	7	6	6				
Delivery &								

Production is planted to begin of October and continue until July 22, 2004

One Module ships to LAT Integration Site (SLAC) every two weeks beginning with FM Apa 17,92004 103 93 48 46
- Two Modules every two weeks for modules 7 thru 16

- During full production, up to ten modules will be in process at once
 - All assembly and test activities will operate in parallel
 - Electronics integration will be capable of handling several modules simultaneously



Assembly and Test Schedule

- Level 3 Milestones Delivery for LAT Integration
 - CAL Flight Modules move through Assembly and Test in groups of 2
 - Modules are Thermal-Vacuum tested in pairs to save \$\$\$
 - Instrument integration schedule specifies required Ready For Integration (RFI) dates
 - RFI rate:
 - FMA, FMB, FM1-FM6
 - One Module every two weeks
 - FM7-FM16
 - Two Modules every two weeks

Module	Planned Module Delivery Date	LAT Schedule Integration Date		
Qual Model (FM A)	17 Feb 04	17 Feb 04		
Flight Spare (FM B)	17 Feb 04	17 Feb 04		
Flight Model 1	5 Mar 04	15 Mar 04		
Flight Model 2	8 Mar 04	15 Mar 04		
Flight Model 3	30 Mar 04	15 Jun 04		
Flight Model 4	30 Mar 04	15 Jun 04		
Flight Model 5	28 Apr 04	15 Jul 04		
Flight Model 6	28 Apr 04	15 Jul 04		
Flight Model 7	25 May 04	29 Jul 04		
Flight Model 8	25 May 04	29 Jul 04		
Flight Model 9	9 Jun 04	12 Aug 04		
Flight Model 10	9 Jun 04	12 Aug 04		
Flight Model 11	22 Jun 04	26 Aug 04		
Flight Model 12	22 Jun 04	26 Aug 04		
Flight Model 13	7 Jul 04	10 Sep 04		
Flight Model 14	7 Jul 04	10 Sep 04		
Flight Model 15	22 Jul 04	24 Sep 04		
Flight Model 16	Naval Rese	arch 522 04		



Flight Module 3 & 4 – Assembly & **Test**

W B S	A c tivity	A c tivity	Total	0 riq	E a rly	E a rly		
M D 3	ID	D e s c rip tio n	Float	Dur	S tart	F in is h	FY 03	FY04 OCT NOV DEC JAN FEB MAR APR MAY JUN JUL AUG SEP O
CAL FM 3								
4.1.5.9 CALORIM 4.1.5.9.2.5	ETER MODULE ASSE				1.0 (0.0 (0.0	1.0.00.00	-	
	5 C 12 19	Receipt in spection	5 6		12/08/03	12/08/03	4	Receipt inspection
4.1.5.9.2.5	5 C 12 2 3	PEM-to-AFEE integration	4 4		01/05/04	0 1 /1 2 /0 4	-	PEM-to-A FEE in tegration
4.1.5.9.2.5	5 C 1 2 2 3 B	TEM & PS Functional Check	6 2		01/13/04	0 1 /1 4 /0 4		TEM & PS Functional Check
4.1.5.9.2.5	5 C 1 2 2 4	TEM & PS integration	6 2		0 1 /1 5 /0 4	01/15/04		TEM & PS integration
4.1.5.9.2.5	5 C 1 2 2 5	Comprehensive State Functional test	6 2		01/16/04	0 1 /2 0 /0 4		Comprehensive State Functional test
4.1.5.9.2.5	5 C 1 2 2 6	Electronic calib	6 2		0 1 /2 1 /0 4	0 1 /2 2 /0 4		E lectronic calib
4.1.5.9.2.5	5 C 1 2 2 7	Muon calibration #2	6 2	2	0 1 /2 3 /0 4	01/26/04		Muon calibration #2
4.1.5.9.2.5	5 C 1 2 2 8	Mass properties #2	6 2	1	0 1 /2 7 /0 4	01/27/04		Mass properties #2
4.1.5.9.2.5	5 C 1 2 3 0	Vibration test	6 2	3	0 1 /2 8 /0 4	01/30/04		D+V ib ration test
4.1.5.9.2.5	5 C 1 2 3 1	M uon calibration #3	5 4	2	03/01/04	03/02/04		Muon calibration #3
4.1.5.9.2.5	5 C 1 2 3 2	Comprehensive Functional Test #2	5 4	2	03/03/04	03/04/04		Comprehensive Functional Test #2
4.1.5.9.2.5	5 C 1 2 3 3	Preship review and signoff	5 4	1	03/05/04	03/05/04	1	Preship review and sign off
4.1.5.9.2.5	5 C 1 2 3 4	m argin	5 4	5	03/08/04	03/12/04	1	Impara pg in
4.1.5.9.2.5	5 C 1 2 3 4 A	Ship to SLAC	5 4	7	0 3 /1 5 /0 4	0 3 /2 3 /0 4	1	ship to SLAC
4.1.5.9.2.5	5 C 1 2 3 4 B	Post Ship Functional	5 4	5	0 3 /2 4 /0 4	03/30/04	1	Post Ship Functional
4.1.5.9.2.5	5 C 1 2 3 5	AV: Flight Calorimeter Tower 3 RFI	5 4	0		03/30/04		AV: Flight Calorim eter Tower 3 RFI
CALFM4								
	ETER MODULE ASSE				T		-	• •
4.1.5.9.2.6	5 C 1 2 3 9	Receipt in spection	6 2		12/15/03	12/15/03	4	Receipt inspection
4.1.5.9.2.6	5 C 1 2 4 3	PEM-to-AFEE integration	4 4		01/21/04	0 1 /2 8 /0 4		PEM-to-AFEE integration
4.1.5.9.2.6	5 C 1 2 4 3 B	TEM & PS Functional Check	5 3		01/29/04	0 1 /3 0 /0 4		iHTÉM & PS Functional Check ▼ II
4.1.5.9.2.6	5 C 1 2 4 4	TEM & PS integration	5 3		0 2 /0 2 /0 4	02/02/04		TEM & PS integration
4.1.5.9.2.6	5 C 1 2 4 5	Comprehensive State Functional test	5 3		0 2 /0 3 /0 4	02/05/04		Comprehensive State Functional test
4.1.5.9.2.6	5 C 1 2 4 6	Electronic calib	5 3	2	02/06/04	02/09/04		Electronic calib
4.1.5.9.2.6	5 C 1 2 4 7	Muon calibration #2	5 3	2	02/10/04	02/11/04		Muon calibration #2
4.1.5.9.2.6	5 C 1 2 4 8	Mass properties #2	5 3	1	0 2 /1 2 /0 4	02/12/04		Mass properties #2
4.1.5.9.2.6	5 C 1 2 5 0	Vibration test	5 3	3	0 2 /1 3 /0 4	02/18/04		Vibration test
4.1.5.9.2.6	5 C 1 2 4 9	3 & 4 Thermal vac functional test	7 5	10	02/19/04	02/28/04		3 & 4 thermal vac functional test
4.1.5.9.2.6	5 C 1 2 5 1	Muon calibration #3	5 4	2	03/01/04	03/02/04		Muon calibration #3
4.1.5.9.2.6	5 C 1 2 5 2	Comprehensive Functional Test #2	5 4	2	03/03/04	03/04/04		Comprehensive Functional Test #2
4.1.5.9.2.6	5 C 1 2 5 3	Preship review and signoff	5 4	1	0 3 /0 5 /0 4	03/05/04		hPreship review and sign off
4.1.5.9.2.6	5 C 1 2 5 4	m argin	5 4	5	03/08/04	03/12/04]	t a gin a gin
4.1.5.9.2.6	5 C 1 2 5 4 A	Ship to SLAC	5 4	7	03/15/04	0 3 /2 3 /0 4]	Ship to SLAC
4.1.5.9.2.6	5 C 1 2 5 4 B	Post Ship Functional	5 4	5	0 3 /2 4 /0 4	0 3 /3 0 /0 4		Post Ship Functional
4.1.5.9.2.6	5 C 1 2 5 5	AV: Flight Calorim eter Tower 4 RFI	5 4	0		0 3 /3 0 /0 4	1	AV: Flight Calorim eter Tower 4 RFI





Facilities

Assembly and test facilities

- **Naval Center for Space Technology**
- NRL Code 8000, Building A59
- Assembly in 1000 sq. ft. clean room
 - Additional space available as needed
- **Environmental test**

 NCST facilities and experienced operators



Thermal Vacuum test facility

Vibration test facility

